N121M Performance Test. 5/16/2021 Jim Hudson, PIC. Gordon Hall CO-Pilot, Logan Schwisow - pax

The objective of the test was to see how she flies, and performs in stalls, take-offs and landings and how the numbers compare to the Robertson STOL-STC where applicable.

Conditions: We were within 15# of gross weight and almost neutral CG (see attached W&B sheet). I was PIC, Gordon co-pilot, and Logan riding in the back.

Surface Conditions Wind Calm, Temp 16 deg C, Press 29.92 (for real) near standard conditions. I used the CloudAhoy app to get the take-off/landing distances.

https://www.cloudahoy.com/debrief/?key=V9XpbgSPHLe24Np446po&startAt=1621177635418.02

1st – take-off and pattern at normal C182 numbers – as per the checklist.
0 Flaps, Rotate at ~ 50, climb out at Vy 75 and 775 '/min.
Rotation at 972' Clr 50' 1960'
Standard C182Q take off table says 880 / 1705 for short field take-off with 20 deg flaps. The above was not short field.

Pattern			
Downwind	85Kts	15″ MP	0 Flaps level flight
Initial descent	80 kts	12" MP	10° flaps – 500'/min
Base	70 kts	13" MP	20° flaps – 500'/min
Final	60 kts	13" MP	40° flaps – 500'/min, 55kts on short final

Touch and go – rotate at 55 climb out 80kts = 600 '/min. Went out to the practice area to do stall tests.

Slow Flight – MCA/Stall 6000' 29.92 Press OAT= 58° F Aircraft Weight: 2900#

Slow Flight/Stall Test: Determine the power to maintain MCA (Minimum Control Airspeed) in level flight. – Stall Horn Just Starting to sound. Reduce Power while maintain altitude until Stall. Record IAS for each configuration.

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Slow	Flight	/ Stall

Flaps	IAS @ MCA	IAS @ Vs	MCA-Vs	PWR MP/RPM	STC Stall Speeds	POH Stall Speeds	Notes
0	55	40	15	15 /2350			
20	45	30	15	16 /2350			
40	45	30	15	16 /2350			
0	57	53	4	Pwr Off	44	45	0° bank
20	48	45	3	Pwr Off	37	43	0° bank
40	46	<mark>38</mark>	8	Pwr Off	<mark>36</mark>	42	0° bank
0	60	55	5	Pwr Off	50		20° bank
0	73	<mark>65</mark>	8	Pwr Off	<mark>64</mark>	54	45° bank
20	50	48	2	Pwr Off	40		20° bank
20	55	55	0	Pwr Off	51	50	45° bank

The result of this test is that the stall speeds we achieved in the test were quite a bit higher that the POH, except at 40° Flaps, no bank, No flaps 45° bank 0. The STC does not indicate the altitude at which the stall tests were performed, nor does the POH. The STC gives numbers for 40 deg bank angle instead of 45 deg as the POH. The POH states that KIAS values are approximate. This is why it's best to go out and test the actual plane. Lori MacNichol has all of her students do this, because each plane is different, even the same model. To be more statistically correct, we should perform this test several times.

We did some power on Stalls but did not record the numbers. I do recall the stall speed being below the minimum indicted speed of 40 on the stall speed indicator – I estimate lower than 30kts.

Best Glide POH says 70, STC says 69. We tested at 70 and 65. 70 Kt = -750'/min 65 Kt = -700'/min with prop out 650'/min. Seems like 70 is a good number.

Back for landings.

1st landing T&G about the same as above.

2nd Short field landing – estimated landed in 700' CloudAhoy says 642'.

Short Field take/off 20° Flaps as per stc – rotated at 45kts. I don't recall the take-off roll, I think less than 1,000'. We started at the 1,000' mark. One thing I noticed is when retracting the flaps, there was significantly more drop than in our regular C182's. Something to be aware of with the R-STOL. The stalls seemed a little milder than a regular C182. I did the falling leaf without power and it did break and recover similar to a normal C182s, but possibly a bit milder. Gordon and I both feel that 55 Kts on final is about as low as one should go to not have the bottom drop out. 60 Kts on final felt very stable. There was no wind to really test the lack of crosswind authority with the R-STOL.

Check Out Requirements.

Ground check-out that reviews inspection of the VG's, operation of the Electric Tach, and Engine monitor, operation of Robertson STOL, review of the STC, and general familiarization with the plane. Flight Check-out – Stalls in various configurations and several take-off and landings at normal and short field configurations.

Attached is the data sheet Gordon and I used and pdf's. Cheers,

Jim

Basic Information					
Aircraft Ident: N121M	Aircraft Type: C-182Q	Departure Date:	Departure Time:	9:00 Arrival Time	12:00
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Fuel - 75 Gallons MAX Useable	75.0	-
Planned Trip Time	1.5	Hrs.
Payload (Pax & Baggage)	655.0	
* Range @ 74% Pwr =12.7 GPH	5.8	Hrs.
Fuel Reserve Time	4.3	Hrs.
* Desce beesd as DOH First Division	2 1 1 0/ C	0 0

Enter data in highlighted blocks			
Max Gross Weight	2950		
Take-Off Weight	2928	%66	of Gross
Over/Under weight	22		

Range based on POH Fuel Burn @ 74% power, 8,000' Std Conditions - may be more or less depending on leaning, DA, other factors.

weight and balance at bepartury	,			
Loads	Weight		Arm	Moment
(P	ounds)		(Inches)	/1000
Empty Aircraft: 18	831.8		37.26	68.26
Front Passengers:	200.0	195.0	37.0	14.6
Rear Passengers:	225.0	15.0	74.0	17.8
Area 1 Baggage 120# Max:			0'.70	
Area 2 Baggage 80# Max:	20.0		115.0	2.3
Departing Fuel : 73.5 4	441.0		47.9	21.1
Grnd Ops (Gal): 1.5				
Totals: 29	927.8		42.4	124.1
CG = Tota	al Momei	nt / Total	Weight:	42.4



59	60	60) Deg Flaps	SF Lnd @ 4(
47	48	48	VS1	
44	45	45	V S0	
68	70	70	V BG	
108	111	111	Va	
Wt.	Off Wt	Wt		
At Land	At Take	At Gross		

$\begin{tabular}{ c c c c c c c c c c c c c c c c c c c$	Weight and Balance at Arr	ival		
(Inches) (Inches) /1000 Empty Aircraft: 1831.8 37.26 68.26 Front Passengers: 395.0 37.0 14.6 Rear Passengers: 240.0 74.0 17.8 Baggage (Area 1): 240.0 74.0 17.8 Baggage (Area 1): 240.0 115.0 2.3 Arrival Fuel (Gal) 54.5 326.7 47.9 15.7 Arrival Fuel (Gal) 54.5 328.7 47.9 15.7 Arrival Fuel (Gal) 54.5 328.7 47.9 15.7 CG = Total Moment / Total Weight: 22.3 118.6 23.2	Loads	Weight	Arm	Moment
Empty Aircraft: 1831.8 37.26 68.26 Front Passengers: 395.0 37.0 14.6 Rear Passengers: 240.0 74.0 17.8 Baggage (Area 1): 240.0 74.0 17.8 Baggage (Area 1): 97.0 2.3 37.26 Arrival Fuel (Gal) 54.5 326.7 47.9 15.7 Arrival Fuel (Gal) 54.5 326.7 47.9 15.7 CG = Total Moment / Total Weight: 42.2 118.6		(Pounds)	(Inches)	/1000
Front Passengers: 395.0 37.0 14.6 Rear Passengers: 240.0 74.0 17.8 Baggage (Area 1): 97.0 17.8 Baggage (Area 1): 20.0 115.0 2.3 Baggage (Area 2): 20.0 115.0 2.3 Arrival Fuel (Gal) 54.5 326.7 47.9 15.7 Arrival Fuel (Gal) 54.5 326.7 47.9 15.7 CG = Total Moment / Total Weight: 42.2 118.6	Empty Aircraft:	1831.8	37.26	68.26
Rear Passengers: 240.0 74.0 17.8 Baggage (Area 1): 97.0 17.8 Baggage (Area 2): 20.0 115.0 2.3 Arrival Fuel (Gal) 54.5 326.7 47.9 15.7 Arrival Fuel (Gal) 54.5 2813.5 42.2 118.6 CG = Total Moment / Total Weight: 42.2 118.6	Front Passengers:	395.0	37.0	14.6
Baggage (Area 1): 97.0 Baggage (Area 2): 20.0 115.0 2.3 Arrival Fuel (Gal) 54.5 326.7 47.9 15.7 Arrival Fuel (Gal) 54.5 326.7 47.9 15.7 Cold Call Totals: 2813.5 42.2 118.6 CG = Total Moment / Total Weight: 42.2 118.6	Rear Passengers:	240.0	74.0	17.8
Baggage (Area 2): 20.0 115.0 2.3 Arrival Fuel (Gal) 54.5 326.7 47.9 15.7 Totals: 2813.5 42.2 118.6 CG = Total Moment / Total Weight: 42.2 118.6	Baggage (Area 1):		97.0	
Arrival Fuel (Gal) 54.5 326.7 47.9 15.7 Totals: 2813.5 42.2 118.6 CG = Total Moment / Total Weight: 42.2	Baggage (Area 2):	20.0	115.0	2.3
Totals: 2813.5 42.2 118.6 CG = Total Moment / Total Weight: 42.2	Arrival Fuel (Gal) 54.5	326.7	47.9	15.7
CG = Total Moment / Total Weight: 42.2	Totals:	2813.5	42.2	118.6
	CG = Total Mo	ment / Total	Weight:	42.2



		AL LANG O	II WEIGIIL			
	Sea @ GW	Sea	2,500	5,000	7,500	10,000
٧x	25	57	58	59	60	61
Vy	78	78	76	75	74	73
		At Landinç	g Weight			
٧x	25	56	57	58	59	60
Vy	78	76	75	74	72	71

PERFORMANCE TEST

DATES/16/21 AIRCRAFT N 121M MODEL 182 Q PILOT Jun / Govden / Logan CONDITIONS: ALTITUDE 000 PRES 29 92 OAT 58 AIRCARFT WEIGHT 2900

Slow Flight/Stall Test: Determine the power to maintain MCA (Minimum Control Airspeed) in level flight. - Stall Horn Just Starting to sound. Reduce Power while maintain altitude until Stall. Record IAS for each configuration.

Slow Flight / Stall

Flans	IAS @ MCA	IAS @ Vs	PWR MP/RPM	Notes	LET DY MALE	
0	45 e Mich	40	15"/2350		11 A 1941	- 0
0	.51	53	Pwr Off		(pr) (2)	-0
10		5	0 / 01		4.51.011	
10			Pwr Off		Polys Tr	
20	45	30	16" 12350		200	
20	48	45	Pwr Off	1 (0 8	0.8	15
30 /	ne datio actin	Base ulider o	/	The second second	10 M	Ū.
30		Do te marcial	Pwr Off		ADS: D	1.1
40	45	30	16" 12350	Prist		- 11
40	46	38	Pwr Off	1 MR 1.	7305	124
0	10	55	Pwr Off	20° bank		10
0	73	65	Pwr Off	45° bank	5,0a	20
20	50	48	Pwr Off	20° bank	20,62	0
20	555	55	Pwr Off	45° bank		

Power On Climb / Stall Full Power Climb at Vx/Vy pitch back until Stall Horn, then Stall. Record Luse NT 2535 2 UPA MUNDE CAIM 70 01 NT

AS and R	OC	IEMIP	loc			61.10 1.001	
Flaps	POH/STC	IAS /	IAS /	PWR MP/RPM	ROC		
	IAS	MCA	VS			D. DOLLANDALA off	1
0 Vr	50/	n/a	14	/	n/a	Per POH std take-off	_
0 Vx	58 /			/		@ 5,000' per POH	
0 Vv	75 / 75			/		@ 5,000' per POH	100/410
20.Vr	50/44	n/a		1	n/a	Short Field t/o	TAKES
20 Vx	57/50		1.11	1	10 10	`@ test alt	
20 Vy	70/68			/		`@ test alt	
0.Vx	58 /			1		20° bank	
20 Vx	75 / 75			1		20° bank	
0.1/x	57/50		1.11	1	2	45° bank	
20.1/2	70/68			/		45° bank	

ZOF

ROTATE 45K 22" ZO"F 300'9 Full Pun

Climb Box Fourfour

600' Full Ror

Cruse / Slow Cruse / Landing

Pattern / Power Off / Landing. Power required to achieve – 500 ft/min Descent.

Flaps	POH/STC IAS/GPH	ACTUAL IAS /GPH	PWR MP/RPM	ROC	ora Flight / Stall
0	143 / 13.3	/	22/2400	0	Cruse 6000' / GPH
0	129 / 10.4	/	20/2200	0	Cruse 6000' / GPH
0	116/8.6	/	18/2100	0	Cruse 6000' / GPH
	POH/STC IAS		1000	HO swith "Note: CH	C74 2 2 0
20	80	80	/	0	Slow Cruse
0	70/69		Pwr Off	-	Best Glide (cruse pitch attitude)
0	90/		/	HO we	Downwind
10	80/		12"1	-500	Initial descent
20	70		13"/	-500	Base
40	60-70/ 3	5	14*1	-500	Final
40	60 /57		/	PWI LIF	Final Short Field
0	70 /69		Pwr Off	PWEDE	Initial descent (best glide)
20	1		Pwr Off	10.04	Base
40	65/60		Pwr Off		Final

-700 1273.9 DOWN WHIND 15" = 85K OFT. DESCENT 2250 PR4

DRATIAL 12" = 80x 10°F 500 Fr. 1

BASE 13" = 7010 20°F 500 1 FINAL 19" - 5512 20°F TO TOUCHDOWN